

A comparative analysis of resistance models for austenitic stainless-steel girders subjected to concentrated loads

Análisis comparativo de modelos de resistencia para vigas esbeltas de acero inoxidable austenítico sometidas a cargas concentradas

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Abstract

The increasing use of stainless steel in construction has led to the need of developing resistance models for structural elements made of this material. Unlike carbon steels, stainless steel alloys exhibit stress-strain curves with a pronounced strain hardening capacity and reasonable ductility that should be considered in the design. This difference in behavior makes the formulations used for carbon steel conservative when designing with stainless steel. Therefore, this paper presents a comparative analysis of resistance models for slender austenitic stainless-steel beams subject to concentrated loads. First, the failure mechanisms of stainless-steel beams are presented using a nonlinear finite element model. From this validated numerical model, a database obtained from a parametric analysis that covers a wide range of geometries is presented. Subsequently, this database is used to perform a comparison between various resistance models available in the literature. These models correspond to both international design codes and models obtained through machine learning. Finally, the numerical results show considerable improvement in the predicted ultimate resistances for slender stainless steel plate girders subjected to patch loading.

Keywords: ultimate resistance; finite element; nonlinear analysis; stainless steel; patch loading.

Resumen

El creciente uso de acero inoxidable en la construcción ha creado la necesidad de desarrollar modelos de resistencia para elementos estructurales fabricados con este material. A diferencia de los aceros de carbono, las aleaciones de acero inoxidable exhiben curvas esfuerzo-deformación con una pronunciada capacidad de endurecimiento por

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deformación y ductilidad razonable que deben ser consideradas en el diseño. Esta diferencia en el comportamiento hace que las formulaciones utilizadas para acero al carbono sean conservadoras al momento de diseñar con acero inoxidable. Por lo tanto, este trabajo presenta un análisis comparativo de modelos de resistencia para vigas esbeltas de acero inoxidable sometidas a cargas concentradas. En primer lugar, se presentan los mecanismos de falla de vigas de acero inoxidable utilizando un modelo no lineal por elementos finitos. A partir de este modelo numérico validado se presenta una base de datos obtenida a partir de un análisis paramétrico que cubre un amplio rango de geometrías. Posteriormente, esta base de datos se utiliza para realizar una comparación entre varios modelos de resistencia disponibles en la literatura. Estos modelos corresponden tanto a códigos de diseño internacionales como a modelos obtenidos a través de aprendizaje de máquina (*machine learning*). Finalmente, los resultados muestran una mejora considerable en la predicción de la resistencia última de las vigas esbeltas de acero inoxidable sometidas a cargas concentradas.

Palabras clave: resistencia última; elementos finitos; análisis no lineal; acero inoxidable; carga concentrada.

1. Introduction

At present, the current generation of international design specifications for metal structures (stainless steel and aluminum) [1], [2], [3] have been developed based on carbon steel guidelines [4], [5]. Stainless steel and aluminum exhibit nonlinear material strain-stress curves with significant strain hardening and reasonable ductility. Several research projects [6], [7], [8], [9], [10], [11], [12], indicate that for these metal alloys their strain hardening capabilities should be considered for strength calculations to achieve economical and efficient structural designs. Figure 1 depicts the notation used herein for stainless-steel plate girders subjected to patch loading.



Figure 1. Notation of a stainless-steel plate girder subjected to patch loading.

In the American [1] and European standards [2] for the design of stainless-steel members, the formulation to calculate the resistance of stainless-steel plate girders subjected to patch loading are derived from those corresponding to carbon steel plate girders [4], [5]. The formulation found in the EC3 Part 1-5 [5] is based upon a plastic collapse mechanism developed for carbon steel girders [13]. In consequence, the difference in the stress-strain behavior between stainless steel and carbon steel is neglected.

The resistance of stainless-steel girders subjected to patch loading has been investigated experimentally and numerically [14], [15]. After comparing the experimental and numerical results with theoretical predictions obtained with international standards for carbon steel [4], [5], the results showed that the predicted resistances were generally conservative, especially for stocky girders, i.e girders with low web slenderness ratios.

Recently, Graciano et al. [16] and dos Santos and Gardner [17] calibrated the resistance model for patch loading employed for carbon steel in the EC3 Part 1-5 [5] to predict the corresponding resistance of austenitic stainless steel plate girders. The results showed a significant improvement in the theoretical predictions obtained with the calibrated models.

Finite element analysis (FEA) has been proven to be a reliable tool to predict the behavior of stainless-steel girders subjected to patch loading [14], [15], [16], [17], [18]. To standardize and facilitate FEA, the European Committee for Standardization [19], [20] developed a designation system that abbreviates the features included in various structural analyses. For elastic analysis, the designation system includes (1) linear elastic analysis (LA), (2) geometrically nonlinear analysis (GNA), and (3) geometrically nonlinear analysis with imperfections (GNIA). For plastic analysis, the system includes (1) first-order materially nonlinear analysis (MNA), (2) geometrically and materially nonlinear analysis (GMNA), and the more advanced (3) geometrically and materially nonlinear analysis with imperfections (GMNIA). GMNIA is mostly recommended to account for the influence of material nonlinearity and initial geometric imperfections.

As mentioned above, stainless steel alloys exhibit a stress-strain behavior with larger strain-hardening than carbon steel alloys, which should be taken into consideration in the design.

Therefore, this paper aims at conducting a comparative analysis of resistance models for the design of austenitic stainless-steel girders subjected to patch loading.

The data used for comparison is made of a collection of numerical simulations obtained through GMNIA. These data cover a wide range of geometries. Numerically computed resistances are compared to values predicted using available formulae from the literature. Finally, the numerical results show that there is a considerable improvement in the predicted patch loading resistances of stainless-steel plate girders.

2. Nonlinear finite element analysis

In this section, the response of the stainless-steel girders subjected to patch loading is investigated through GMNIA using the software ANSYS [21]. An extensive parametric study was conducted to evaluate the influence of various geometrical parameters on the patch loading resistance. All components of the plate girder were modeled using the four-node shell element *S181* [21] with six degrees of freedom at each node. To account for material nonlinear behavior, a multilinear true stress-strain curve was defined using Ramberg-Osgood expression Equation (1) with strain hardening.

$$\epsilon = \frac{\sigma}{E} + 0.002 \left(\frac{\sigma}{\sigma_{0.2}}\right)^n \tag{1}$$

Initial geometric imperfections were simulated using the first buckling mode of a plate girder subjected to patch loading obtained through eigenvalue analysis. A maximum geometrical web imperfection of $w_0 = \min(h_w / 100, t_w)$ was considered in the numerical models [2].

Simply supported boundary conditions were employed in the numerical model. In one end, displacements in x and y directions were restricted, and all nodes along the other support were free to move longitudinally (x-direction). A concentrated load was applied on the nodes located over an equivalent length s_s in the upper flange, in which the displacements in the x and z directions and all rotations were restricted, allowing only the vertical displacement in the y direction.

The nonlinear response in the post-buckling region was captured using the Riks method [22]. The finite element model was validated using test results obtained by Unosson [14], see Table 1. Young modulus and Poisson ratio were taken as E=200 GPa and v=0.3.

Figure 2 shows a finite element mesh with 4352 elements selected after performing a convergence study. A good agreement was attained for the F^{Exp}/F_u^{FEA} between experimental and numerical resistances with an average of 0.96.

Specimen	а	h_w	t_w	b _f	t _f	s _s	σ _{0.2w}	$\sigma_{0.2f}$	F ^{Exp}	F_u^{FEA}	F^{Exp}
	[mm]	[mm]	[mm]	[mm]	[mm]	[mm]	[MPa]	[MPa]	[kN]	[kN]	$\overline{F_u^{FEA}}$
Pli 4301:1	998	238	4.10	118.5	11.7	40.0	297	285	176.0	186.8	0.94
Pli 4301:2	996	238	4.10	119.9	11.9	80.0	297	285	196.0	208.0	0.94
Pli 4301:3	1397	316	4.10	121.0	11.9	40.0	297	285	168.0	183.4	0.91
Pli 4301:4	1623	438	4.10	121.2	11.9	40.0	297	285	169.0	179.3	0.94
Pli 4301:5	1682	401	8.80	120.4	12.0	40.0	245	285	478.0	446.2	1.07

 Table 1. Geometric and material properties of the stainless-steel girders tested by Unosson [14]



Figure 2. Finite element mesh.

3. Parametric analysis

In the previous section the numerical model was validated. Using this model, a parametric analysis is performed using the following geometric parameters:

- $h_w = 400, 200 \text{ mm}$
- $a/h_w=1, 1.25, 1.50, 1.75, 2$
- $h_w/t_w=75, 100, 125, 150, 175, 200$
- $s_s/h_w=0.10, 0.15, 0.20, 0.25, 0.30, 0.35, 0.40, 0.50$
- $t_f/t_w=1$, 1.25, 1.50, 1.75, 2, 2.25, 2.50, 2.75, 3, 3.25, 3.50, 3.75, 4

Table 2 shows the remaining dimensions and material properties, that were kept constant through the analysis. It is noteworthy mentioning that the section class for all modeled girders satisfies the requirements for slender Class 4 cross-sections and the flanges meet $b_f/t_f < 25$.

Table 2. Geometric and material properties of the girder

<i>b</i> _f [mm]	E [GPa]	v	σ _{0.2} [MPa]	n	σ _{1.0} [MPa]	σ_u [MPa]
120	200	0.30	285	9	334	611

The patch loading resistance for each combination of the parameters $(h_w, a/h_w, h_w/t_w, s_s/h_w)$, and $t_f/t_w)$ was computed numerically, therefore a total of 5940 runs were performed. The computed resistances are employed in the comparative analysis of the resistance models in Section 5.

Figure 3 illustrates the von Mises stress distribution for various flange-to-web thickness ratios t_f/t_w . For all cases, the most stressed area is distributed in the upper region of the web panel, specifically in the region between the loaded flange and web panel.

The results also show that the most stressed is enlarged when t_f/t_w is increased, and the resistance F_u is enhanced 37.7 %, from 90.3 to 124.3 kN when the thickness ratio t_f/t_w goes from 1 to 4. It is also interesting to observe that the maximum stress at failure in Figure 3 is 1.12 times higher than the yield stress reported in Table 2 (i.e. $\sigma_{0.2}$ =285 MPa). Usually, numerical models for carbon steel girders are elaborated considering a perfect elastoplastic material behavior, hence the maximum stress at failure is equal to the material yield strength. As stated in the introduction, stainless steel alloys exhibit a rounded stress-strain curve with significant strain hardening. Consequently, the level of stress at failure should be larger than its corresponding yield stress.

4. Results

4.1. AISC 370 Specifications

In the AISC 370 Specification [1], the patch loading resistance for transversally unstiffened stainless steel plate girders F_{Rd} is calculated with the same formulation used for carbon steel girders:

(a) When the concentrated force is applied at a distance from the member end greater than or equal to d/2, the ultimate strength is calculated as

$$F_{Rd} = 0.80 f_{yw} t_w^2 \left[1 + 3 \left(\frac{s_s}{d} \right) \left(\frac{t_w}{t_f} \right)^{1.5} \right] \sqrt{\frac{E f_{yw} t_f}{t_w}}$$
(2)

(b) When a concentrated force is applied at a distance from the member end less than d/2: For $s_s/d \le 0.2$

$$F_{Rd} = 0.40 f_{yw} t_w^2 \left[1 + 3 \left(\frac{s_s}{d} \right) \left(\frac{t_w}{t_f} \right)^{1.5} \right] \sqrt{\frac{E f_{yw} t_f}{t_w}}$$
(3)

For $s_s / d > 0.2$

$$F_{Rd} = 0.40 f_{yw} t_w^2 \left[1 + \left(\frac{4s_s}{d} - 0.2\right) \left(\frac{t_w}{t_f}\right)^{1.5} \right] \sqrt{\frac{E f_{yw} t_f}{t_w}}$$
(4)

In Eqs. (2) to (4), d corresponds to the girder depth h_w .

4.2. EC3:1-4

In the EC3:1-4 [2], the procedure to calculate the patch loading resistance of stainless-steel plate girders without intermediate stiffeners F_{Rd} is the same developed for carbon steel girders in the EC3:1-5 [11]:

$$F_{Rd} = f_{yw} \, L_{eff} \, t_w / \gamma_{M1} \tag{5}$$

in which f_{yw} is the yield strength of the web, γ_{M1} is a partial safety factor, and L_{eff} is the effective length for resistance to transverse loads

$$L_{eff} = \chi_F \, l_y \tag{6}$$

The reduction factor χ_F due to local buckling is

$$\chi_F = \frac{0.5}{\bar{\lambda}_F} \le 1.0 \tag{7}$$

and the effective load length l_{y} is given by

$$l_y = s_s + 2t_f \left(1 + \sqrt{m_1 + m_2} \right) \le a$$
(8)

where m_1 and m_2 are dimensionless parameters

$$m_1 = f_{yf} b_f / f_{yw} t_w \tag{9a}$$

$$m_2 = 0.02 \left(h_w / t_f \right)^2$$
 (9b)



Figure 3. Von Mises stress distribution for various flange-to-web thickness ratio t_f/t_w . ($h_w = 400, a/h_w = 1.50, h_w/t_w = 200; s_s/h_w = 0.2$).

In Equation (8), m_2 should be taken as zero if $\bar{\lambda}_F < 0.5$ The slenderness parameter $\bar{\lambda}_F$ is obtained from

$$\bar{\lambda}_F = \sqrt{\frac{F_y}{F_{cr}}} \tag{10}$$

The buckling load F_{cr} is

 $F_{\gamma} = f_{\gamma w} t_w l_{\gamma}$

$$F_{cr} = 0.9k_F E t_w^3 / h_w$$
 (12)

$$k_F = 6 + 2\left(\frac{h_w}{a}\right)^2 \tag{13}$$

(11)

 F_y is the yield resistance

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4.3. Calibration of the resistance function χ_F

Recently, a resistance function χ_F has been used for all verifications of structural compressed members within the EC3:1-5 [11].

$$\chi_{\rm F} = \frac{1.0}{\phi + \sqrt{\phi^2 - \bar{\lambda}_{\rm F}}} \le 1 \tag{14}$$

where

$$\phi = \frac{1}{2} \left[1 + \alpha_{F0} \left(\bar{\lambda}_F - \bar{\lambda}_{F0} \right) + \bar{\lambda}_F \right]$$
(15)

Employing a standard evaluation procedure provided in EN 1990 [24], Graciano et al. [16] and dos Santos and Gardner [17] calibrated, the imperfection factor α_{F0} and the plateau length $\bar{\lambda}_{F0}$ for a given level of safety γ_{M1} . Table 3 shows the results for these calibrations.

Table 3. α_{F0} , $\bar{\lambda}_{F0}$ and γ_{M1} calibrated for stainless steel girders

Proposal	α_{F0}	$\bar{\lambda}_{F0}$	γ_{M1}
Graciano et al. [16]	0.40	0.10	1.05
Dos Santos and Gardner [17]	0.60	0.60	1.10

It is worth pointing out that for the effective load length l_y -in Equation (8)- in the calibration conducted by Graciano et al. [16] the best results were attained with $m_2 \neq 0$, and dos Santos and Gardner [17] only considered $m_2=0$.

4.4. Machine learning approach

Graciano et al. [25] based on a machine learning approach proposed a formulation for the patch loading resistance of austenitic slender plate girders

$$F_{Rd} = 25.13 \,\mathrm{x} 10^{-3} f_{yw} t_w^2 \,\xi_g \tag{16}$$

with

$$\xi_g = \left(\frac{t_f}{t_w}\right)^{0.415} \left(\frac{h_w}{a}\right)^{0.211} \left(\frac{s_s}{h_w}\right)^{0.22} \left(\frac{b_f}{t_f}\right)^{0.224}$$
(17)

5. Comparative analysis of resistance models

This section presents a comparative analysis of the resistance models presented. Figure 4 shows a comparison of the mean values for the ratio between the resistances obtained numerically F_u and those obtained using various formulations F_{Rd} . Comparing the mean values, the methodology proposed by Graciano et al. [25] using symbolic regression yields the lowest value m=1.00.



Figure 4. Comparison of mean values for the ratio F_u/F_Rd obtained using various formulations.

In addition, Table 4 shows the basic statistics for the ratio F_u/F_{Rd} . Using AISC 370 Specifications [1] gives a mean m=1.39, with the largest scatter in the results s=0.57 and becoming unsafe in some cases with a minimum min=0.62. Predicted resistances using the EC3:1-4 [2] are on the safe side with m=1.39 and s=0.19. It is worth noting that the design procedures in [1] and [2] correspond to carbon steel girders, and then extended to stainless steel girders without considering the difference in material behavior.

Table 4. Basic statistics for the ratios F_u/F_Rd. obtained using the formulations in [16], [17] and [25]

Model	min	max	т	S	v
[1]	0.62	2.71	1.39	0.57	0.41
[2]	2.02	1.02	1.60	0.19	0.12
[16]	0.99	1.52	1.30	0.09	0.07
[17]	1.06	1.80	1.45	0.18	0.12
[25]	0.84	1.15	1.00	0.03	0.03

An improvement in the predictions is achieved by calibrating the resistance function in Eq. (14). Graciano et al. [16] attained a mean m=1.30 with the lowest scatter s=0.09. Similarly, dos Santos and Gardner [17] attained a mean m=1.45 with the lowest scatter s=0.18. In both calibrations, the plastic yield resistance was obtained using the yield strength of the material. It may be suitable for carbon steel girders for which strain hardening is usually neglected. As seen in Figure 4, stainless steel girders reached a higher stress level at failure, *i.e.* 1.12 times greater than the material yield strength.

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6. Conclusion

This investigation presents a comparative analysis of resistances models for stainless steel plate girders subject to patch loading. A significant improvement in the predicted patch loading resistances is achieved by calibrating the resistance function for stainless steel girders. Further research is required to consider the strain hardening capacity of stainless-steel alloys in the formulation for the plastic yield resistance. From the results obtained herein, an appropriate level of stress, above the material yield strength, may be considered for this plastic yield resistance.

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